



# NASA Exploration Power Distribution Systems

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# Outline



- **Evolution of Human Exploration Power Systems**
- **Spectrum of Power Systems and Voltages**
- **Power Quality**
- **Aerospace Program EPS Development**
  - – **AMPS Modular Power Components**
  - – **AMPS Standardized Modular Power Interfaces**



# Evolution of Human Exploration Power Systems



## EARTH RELIANT

Initial Exploration Missions

### ISS

- Multi-segmented modules  
**>200kW**
- Multiple assembly flights
- Chemical propulsion (reboost)
- Load shed tables, basically manual ops

## PROVING GROUND

Extending Reach

Into The Solar System

### Space Power:

- Multi-segmented modules  
**>100kW**
- Multiple assembly flights
- Electric propulsion dominates
- Manual to semi-autonomous ops

### Surface Power:

- Multi-segmented power units  
**kW to >10kW each**
- Manual to semi-autonomous ops

## EARTH INDEPENDENT

Exploring Other Worlds

Planetary Exploration

### Space Power:

- Multi-segmented modules  
**>300kW to multi-MW**
- Multiple assembly flights
- Electric propulsion dominates
- > Semi-autonomous ops

### Surface Power:

- Multi-segmented power units  
**>10's kW each**
- ≥ Semi-autonomous ops

Near Earth  
Missions

Extended  
Missions

Mars/Moon  
Missions

ISS Deep Space & Mars Risk Reduction

Deep Space Port Mars Preparation

Missions: Months  
Return: hours

Missions: < year  
Return: days

Missions: > years  
Return: > months

# Spectrum of Power Systems and Voltages



EARTH RELIANT

PROVING GROUND

EARTH INDEPENDENT

## Voltage Standards

28Vdc  
120Vdc  
160Vdc

Solar Arrays  
Batteries

Batteries

Solar Arrays  
Batteries  
Fuel Cells  
Surface array  
Surface nuclear  
(Brayton, Stirling)

## Voltage Standards

28Vdc  
120Vdc

120Vdc to >300Vdc (electric propulsion)  
AC distribution for Brayton or Stirling

## Voltage Standards

28Vdc  
120Vdc

>300Vdc (electric propulsion)  
AC distribution for Brayton or Stirling

★ Major issue with Rad Hard Components

Solar Arrays (0.8 – 2AU)

Nuclear  
Batteries  
Flywheels

Super-Capacitors  
Surface Fuel Cells

Surface array  
Surface nuclear  
(Brayton, Stirling)

Complexity and International mix makes it imperative to develop comprehensive power quality standards from the beginning



# What is Power Quality?



## By Wikipedia –

In its broadest sense, power quality is a set of boundaries that allows electrical systems to function in their intended manner without significant loss of performance or life. The term is used to describe electrical power that drives an electrical load and the load's ability to function properly with that electric power. Without the proper power, an electrical device (or load) may malfunction, fail prematurely or not operate at all.

# Power Quality Lessons



- **Historically space power systems requirements have been unique for every platform**
  - Roadblock for re-use of equipment and standardization
- **Majority of the existing power quality specifications were developed for engine driven alternators for aircraft with AC transformers and rectifiers for DC.**
- **Space power systems consist of solar arrays, electrochemical systems, thermal electric generators, etc.**
  - All these sources have unique characteristics
- **Spacecraft loads are often designed for unique platforms**
  - Adapted to meet unique voltage ranges and power quality characteristics
- **Common standard allows transportability to a new platform with high confidence for minimal requalification**
- **Achieve reliability through verification**
  - ISS had remarkable success - Pre-assembled flight elements never tested together were attached to each other for the first time on-orbit and worked



# Characteristics of a Power Quality Standard



- Address both Sources and Loads
- Provide a load and source characteristic for proper operation and protection
- Provide requirement rationale to clarify intent
- Provide verification approach for each requirement
- Provide typical test procedures
- Allow for design flexibility
- Clearly describe interface (connection) power for loads
- Specific specifications can then be developed and coordinated using the Standard

# We now have SAE AS5698 Power Quality Standard



## Standard addresses Power Quality for 28 and 120 Volt DC

### Source Requirements

- Steady State Voltage Range
- Transient Voltage
- Ripple Voltage
- Stability
  - Stability Criteria
  - System Impedance
- Power Isolation
- Common Mode Noise
  - Voltage
  - Current
- Inrush Surges
- Reverse Current
- Fault Clearing
- Fault Coordination

### Load Requirements

- Operating Voltage Range
- System Voltage Transients
- System Ripple Voltage
- Stability
  - Stability Criteria
  - Load Input Impedance
- Input Isolation
- Common Mode Noise
  - Voltage
  - Current
- Limiting Inrush Surges
- Limiting Reverse Current
- Overloads





## Aerospace Program for EPS Development

# Advance Exploration System (AES) Program **AES M**odular **P**ower **S**ystems (**AMPS**) Project



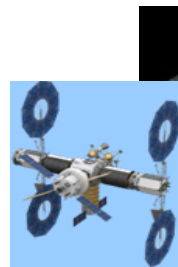
# Advanced Exploration Systems (AES) Program

## **AES** **M**odular **P**ower **S**ystems (**AMPS**) Project

### Objectives:

- Develop modular power units which, when combined with standardized interfaces, can provide commonality across a variety of exploration vehicles.
- Infuse new technology into power systems and components, and prove their capabilities in exploration-based ground demonstrations

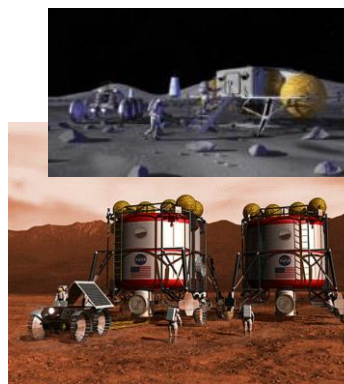
### Applications



Exploration Missions



Mars / Lunar Rovers



Planetary Outposts

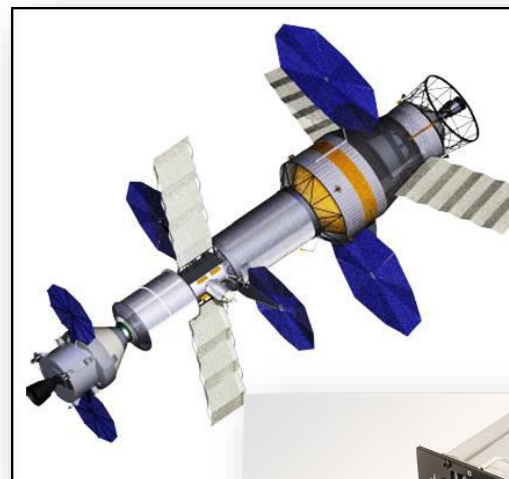


EVA Suits



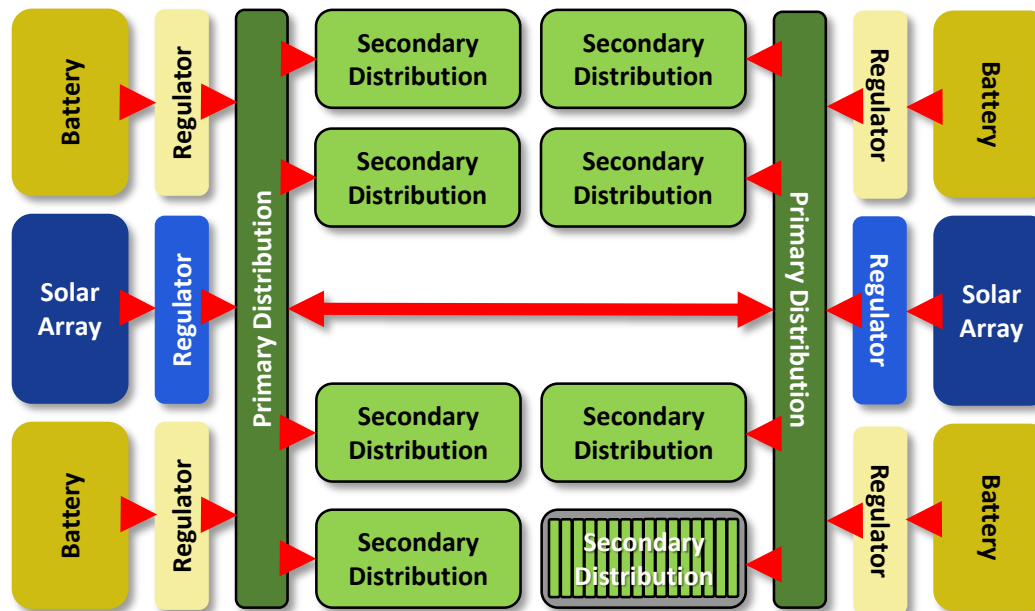
# Why modular?

- Deep Space Missions require lower logistic costs
- Reusability
  - Interchangeability between multiple platforms/missions
  - Reduced spare components needed
  - Scavenging components from retired platforms
- Flexibility
  - Scalable for a wide range of architecture sizes/power levels
  - Spectrum of architectures from centralized to distributed
- Reliability
  - Failures cause reduced capability, not complete loss of use
  - Hot-swap replacement minimizes down-time & system effects



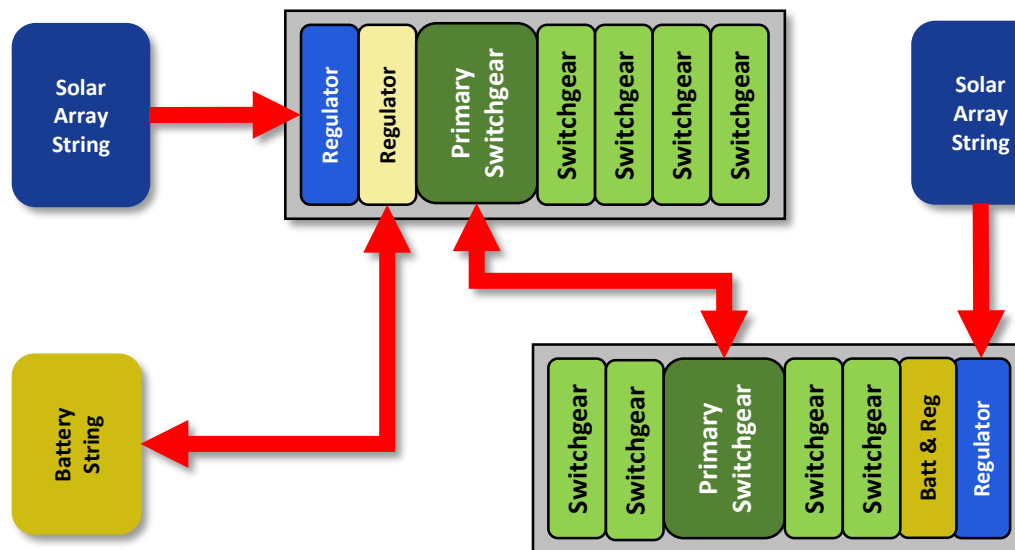


## General Hierarchical Architecture



"Monolithic" units can be replaced by sets of common modules

## Flexible/ Distributed Architecture



PMAD units containing mixed modules enable distributed architectures





# ***Modular Component Development***

- Prototypes will drive development of modular standards.
- Batteries and solar cells are already “modular”, so the next step is to develop modular PMAD components:
- Switchgear features:
  - Remote control
  - Fault containment: current limit and trip
- Converter features:
  - Source/storage/interface/bus regulation
- Common features:
  - Parallelable to grow in power (“tiers” may be required)
  - Digital control, enabling:
    - Intelligent control / automation at the module level
    - Master-less communication for “swarm” intelligence

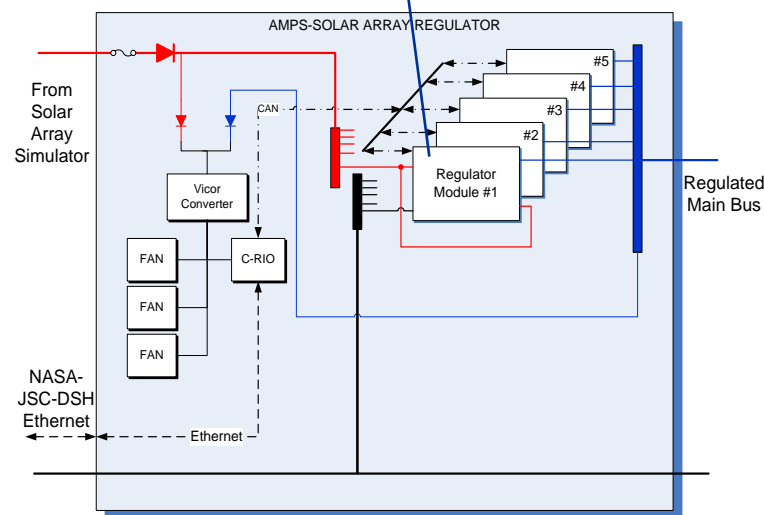
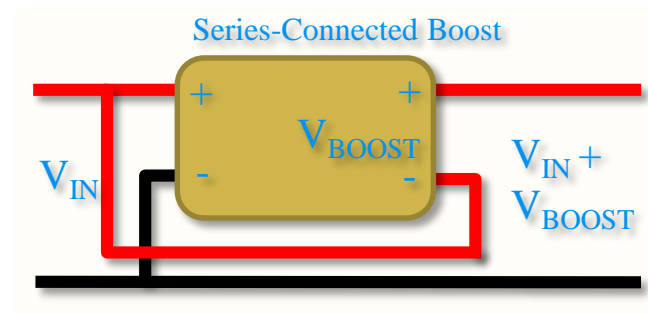
# Solar Array Regulator Development

## Objective:

- Demonstrate 5kW 120V Solar Array Regulator (SAR)

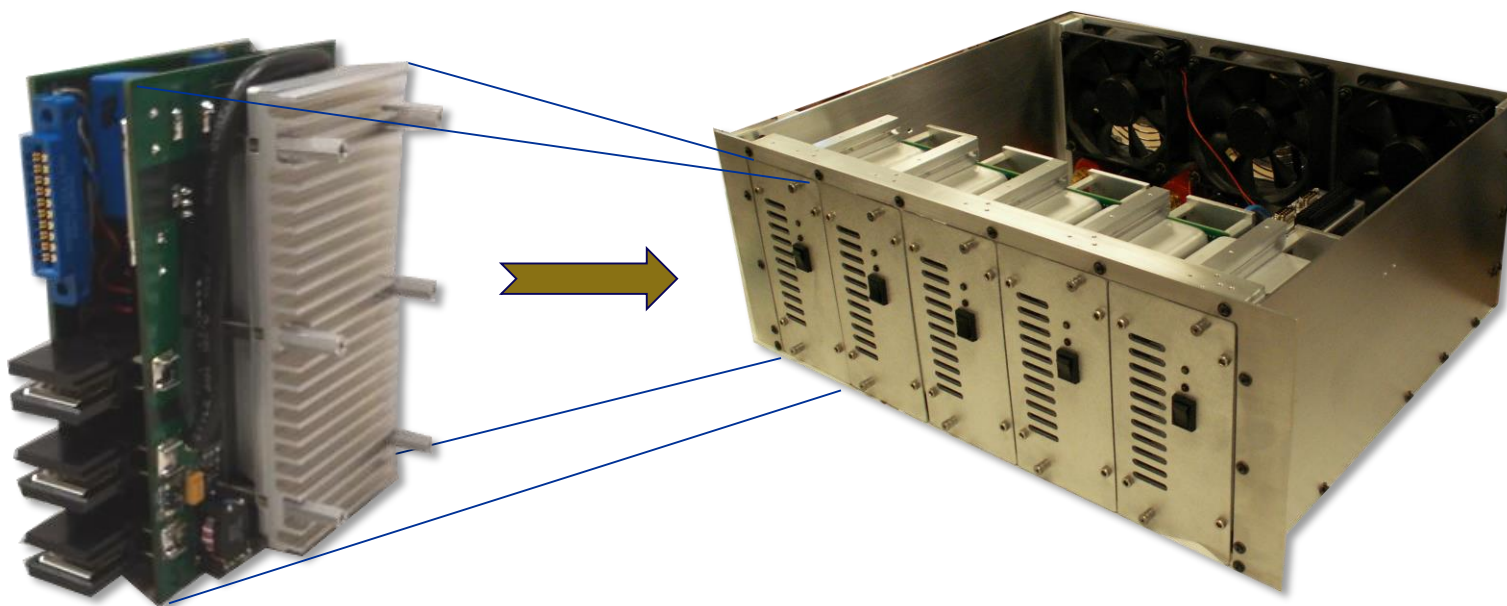
## Current Implementation:

- COTS 120V-to-28V converter
- Series-connected boost configuration: 1kW/module
- Digital control (microcontroller)
  - CAN communication (master-less)
  - Remote control/fault simulation
  - Telemetry
  - Dynamic output voltage set-point (battery charging)
  - Intelligence: efficiency optimization
- Hot-swappable enclosure





# ***Solar Array Regulator Development***



**1 kW Module**

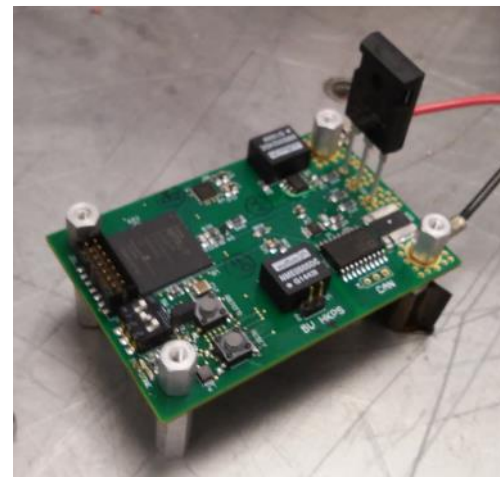
**5 kW Unit**

- 
- **Current Status:**
    - Operating in exploration systems controls lab at JSC
    - Regulates bus voltage and bus-tied battery charge current
  - **Future Work:**
    - Replace power stage with custom bidirectional module
    - Replace microcontroller with FPGA

# Switchgear Development

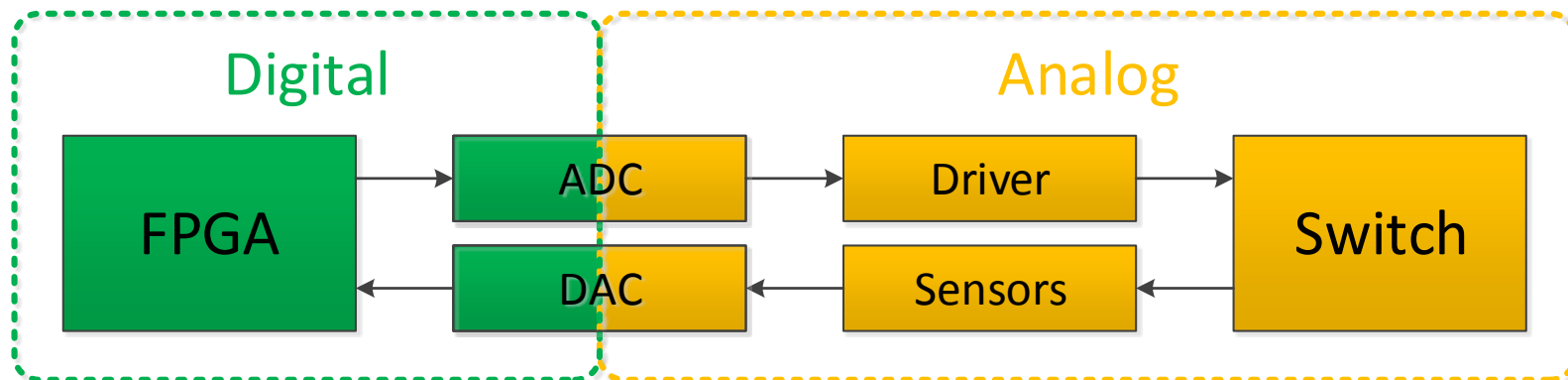
## Objective:

Demonstrate a digitally controlled remote power switch emulating the characteristics of the state-of-the-art



## Implementation:

- 1200V/42A MOSFET & shunt
- ADC (sensor) & DAC (driver)
  - High speed
  - Non-pipelined (low prop. delay)
- FPGA
  - CAN communication
  - Telemetry
  - Current limiting
  - Thermal trip (power-time curve)





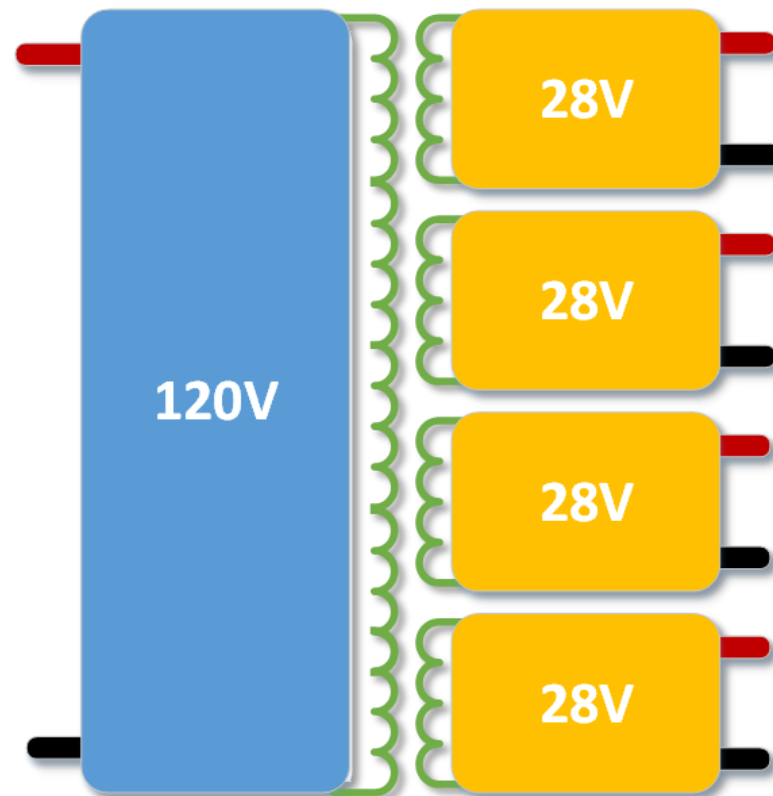
# ***Bidirectional Converter Development***

## Objective:

- Demonstrate a 1kW multifunction bidirectional converter
  - Battery charge/discharge control
  - Solar array regulator
  - Spacecraft interface converter
  - Bus/point-of-load converter

## Implementation:

- 120 V<sub>IN</sub>, configurable output:
  - 4 outputs in series: 8.3A @ 120V
  - 4 outputs in parallel: 35.7A @ 28V
- Modules can be paralleled
- Topology
  - Forward: Full bridge
  - Reverse: Weinberg/flyback
  - Isolated





# ***Modular Power Components Summary***

- The AMPS project is developing and demonstrating modular power system concepts for deep-space human exploration missions.
- Modular power components enable scalable and flexible power system architectures.
- Three prototypes are in development to assist in developing modular power standards:
  - Solar array regulator (5x1kW modules @ 120V)
  - Switchgear (current-limiting to 4A @ 120V)
  - Bidirectional converter (1kW @ 120V/4x28V)



# ***AMPS Standardized Modular Power Interfaces for Future Space Explorations Missions***





# ***AMPS Standardized Modular Power Interfaces***

AMPS is drafting a proposed standard that is:

- Applicable to NASA exploration,
- Accommodates variations in power architecture
- Supports mission flexibility (configuration changes)
- Defines the common infrastructure needed to support the modular design
- Standardizes Data, Electrical and Mechanical Interfaces

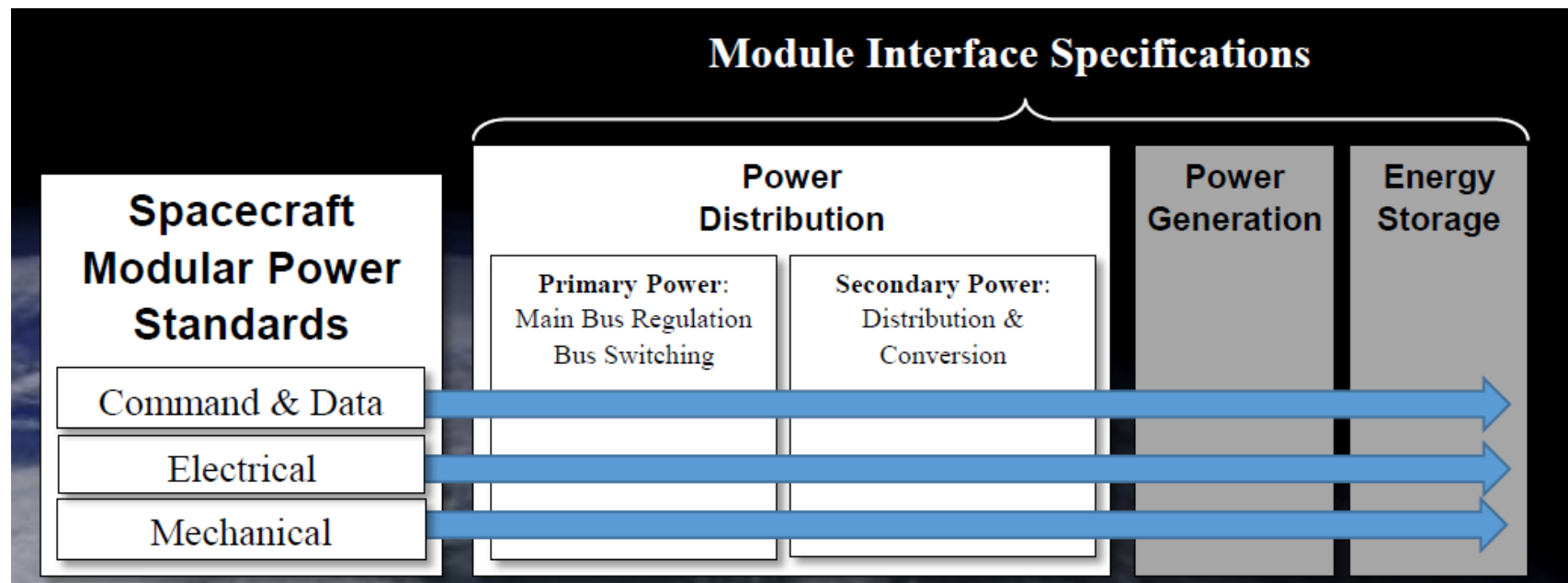
The intent is to guide power system developers without restricting design or technology options.

- Adopts existing standards where applicable
- Emphasize Interchangeability and Interoperability



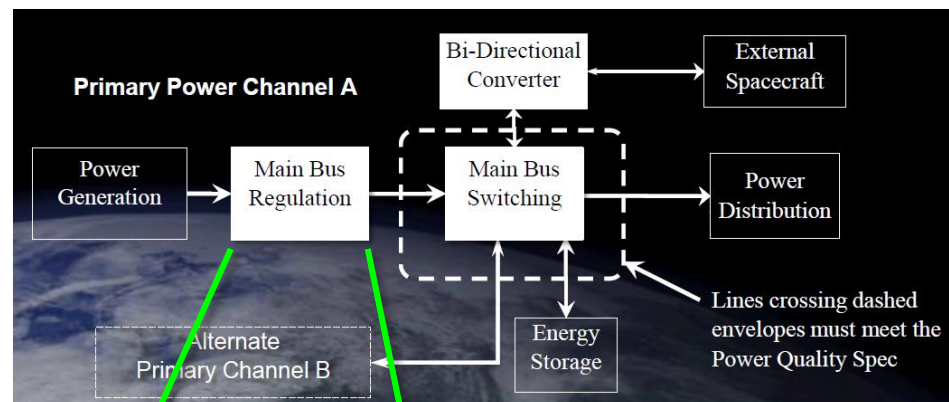
# ***AMPS Standardized Modular Power Interfaces***

- Establish a common framework for Data, Electrical, Mechanical interfaces.
- Apply the Standards to 3 segments of a Power Architecture
- Define interfaces between modules and internal to modules
- Create Interface Specs for
  - Assemblies,
  - Subassemblies
  - Components



# Standardization Frameworks

- Electrical Interface section addresses modular approach that is flexible, configurable, and supportable
  - Breaking an architecture into functional blocks
  - Grouping functions as common modular elements
  - Creating an interconnection framework of Common Backplanes
  - Defining the characteristics that make up Modular Interface Specs

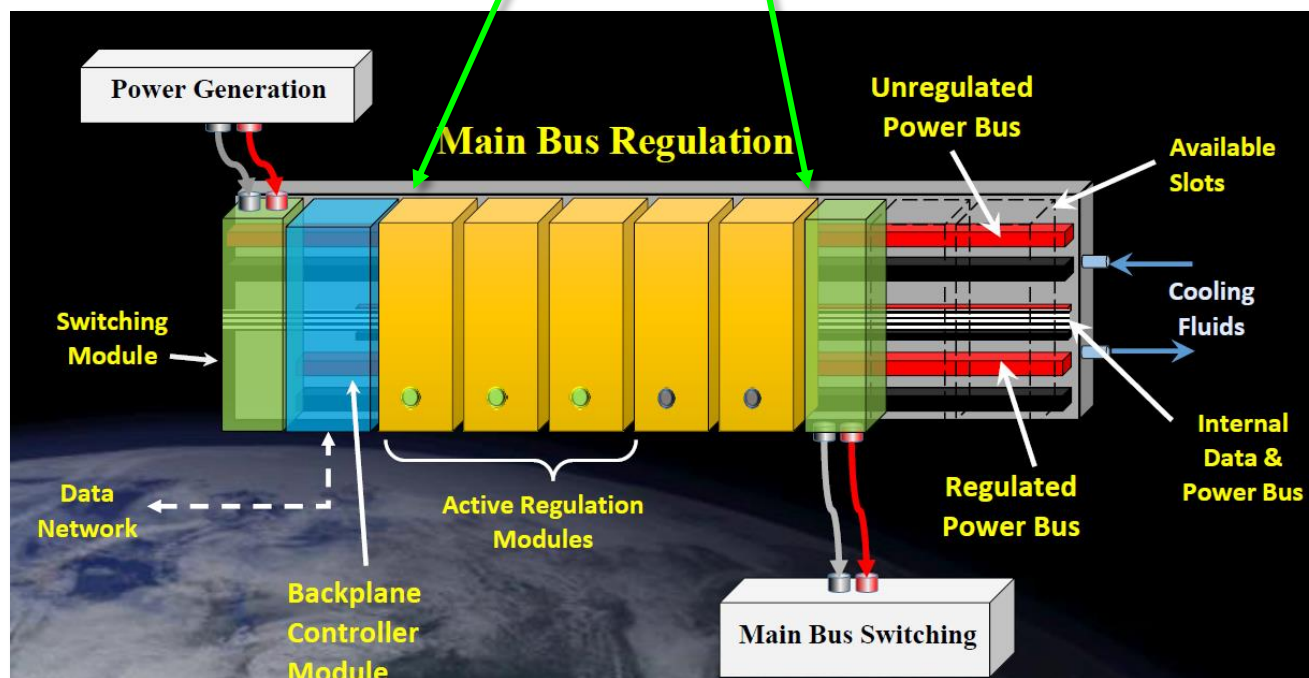


## Primary Power Regulation Backplane-Module

Modules mounted on a Assembly Level Backplane.

Unregulated & Regulated Power, Data and Structural and Thermal Interfaces

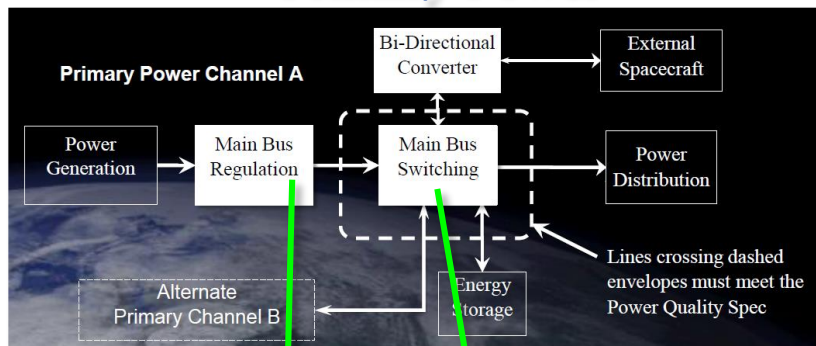
Modules: Switching, Regulation, Unit Control



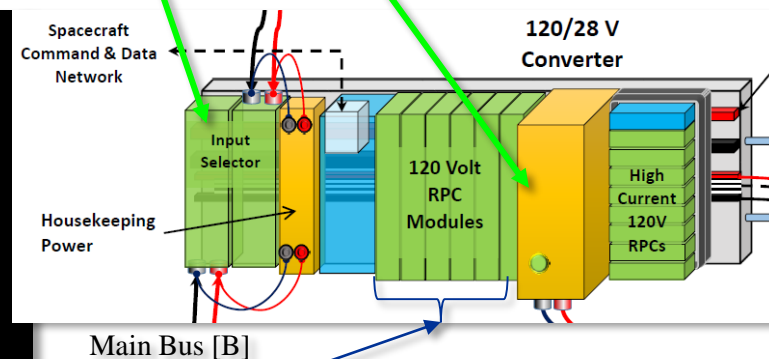
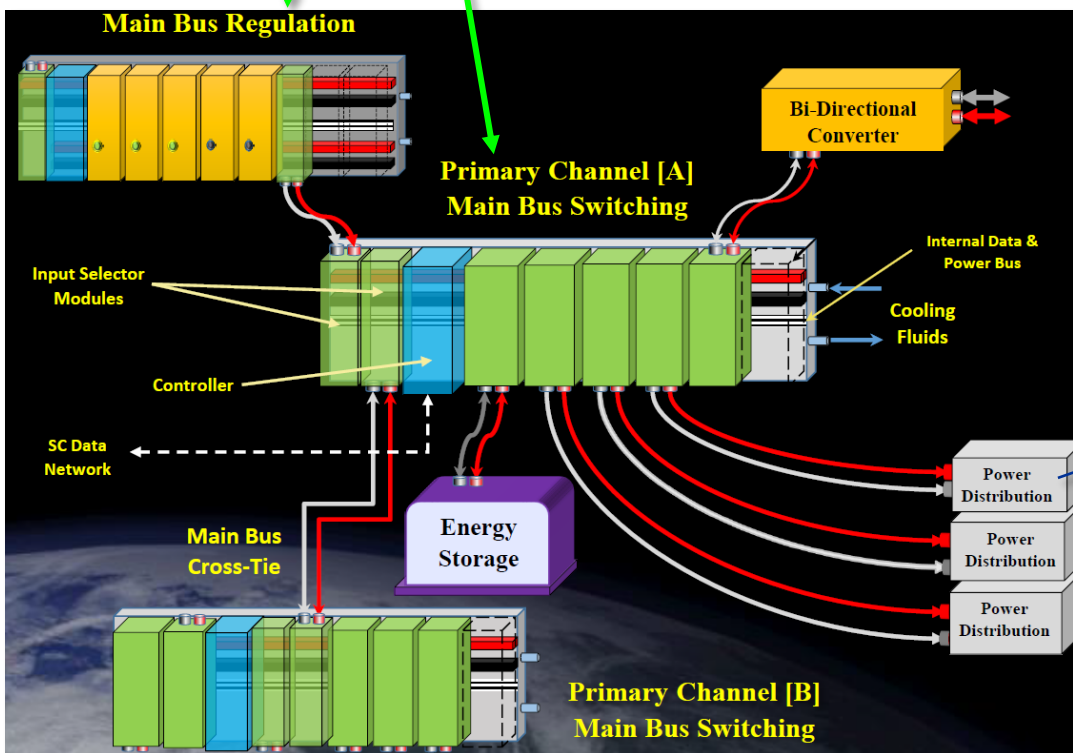
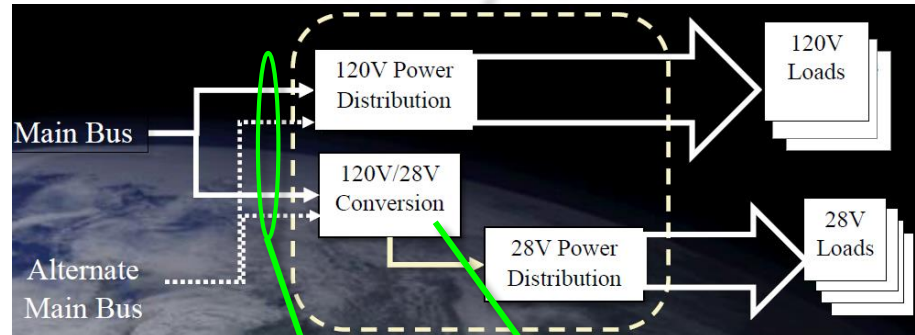


# Power Assembly Backplanes-Modules

## Primary Power



## Secondary Power





# ***Additional Standardization Frameworks***

- Command & Data Interface section addresses the Communication protocols and Software with emphasis on interoperability standards.
  - Power modules will support automatic ID, Digital Configuration and Integration. (i.e. Plug-and-Play)
  - Internally, modules adopt protocols suited power applications but must support the higher level Interoperability requirements.
- Mechanical Interface section addresses the mechanical needs in terms of structural support, encapsulation and thermal control.
  - Modules and backplanes must support static and dynamic loads while providing a means of transferring thermal loads.
  - Mechanical interfaces must assure ease of access and interchangeability.



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